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First-Order Bragg SAR Imagery of Kelvin Wakes  
and Moving Internal Wave Fields

B.A. Hughes

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## DEFENCE RESEARCH ESTABLISHMENT PACIFIC

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DREP Technical Memorandum 89-13

FIRST-ORDER BRAGG SAR IMAGERY OF KELVIN WAKES  
AND MOVING INTERNAL WAVE FIELDS

by

B. A. Hughes

June 1989



Approved by:

  
CHIEF

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Abstract

Two theoretical investigations are given relating to SAR imaging of perturbed surface waves. The first is an analysis of the effect of a long wavelength corrugation on the SAR image formed by 1<sup>st</sup> order Bragg scattering from a deterministic sea wave pattern such as a Kelvin wake. A standard SAR processor is assumed. The second is a stochastic treatment of the SAR imagery formed by a 1<sup>st</sup> order Bragg scattering process from surface waves modulated by the horizontal surface currents of internal waves. Propagational velocities (normalized by the SAR platform velocity) are included to first order. For the Kelvin wake analysis it is shown that the vertical curvature of the corrugating wave can rotate the effective Bragg acceptance angle by up to  $\sim 10^\circ$  for typical wave parameters. For the internal wave analysis, SAR resolutions are given, and in particular it is shown that typical azimuthal velocity components are important in determining the azimuthal resolution dimension of the scene, especially for resolutions of a few metres size.

Résumé

Le texte décrit deux études théoriques sur l'imagerie SAR de vagues de surface perturbées. La première est une analyse de l'effet d'une cannelure de grande longueur d'onde sur l'image SAR formée par la diffusion de Bragg du premier ordre à partir d'un diagramme déterministe des vagues de l'océan, par exemple un sillage Kelvin. On suppose le recours à un processeur SAR standard. La deuxième étude consiste en un traitement stochastique de l'imagerie SAR formée par la diffusion de Bragg du premier ordre de vagues de surface modulées par les courants de surface horizontaux des ondes internes. Les vitesses de propagation (normalisées par la vitesse de la plate-forme SAR) sont données jusqu'au premier ordre. L'analyse de sillage Kelvin permet de démontrer que la courbure verticale de la vague de cannelure peut faire tourner jusqu'à  $10^\circ$  l'angle d'acceptation efficace de Bragg pour des paramètres de vagues typiques. Dans le cas de l'analyse des vagues internes, on précise les valeurs de la résolution SAR, et on constate plus particulièrement que les composantes de vitesse azimutale typiques sont importantes pour déterminer la dimension de résolution azimutale de la scène, en particulier pour les résolutions de l'ordre de quelques mètres.

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## 1. Introduction

Several studies<sup>1,2,3</sup> have been made of the effect of motion in scene elements on SAR processing. In those studies the nature of the motion has been either random<sup>1</sup>, in which case temporal coherence estimates have been given with their effect on the final image, or organized in the form of surface waves<sup>2</sup> but with random phasing relative to the Bragg scatterers. In either case it has been shown that "velocity bunching" takes place, i.e., image misregistrations occur arising from misinterpretations of phase histories due to the combination of platform velocity and scene element velocity. The decorrelation time associated with random motions also serves to increase the effective azimuthal SAR resolution distance<sup>3</sup>. Scene acceleration effects can also be important<sup>2</sup> producing direct changes in resolution.

Two separate problems are discussed in the present paper, SAR imagery of (a) Kelvin wakes, and (b) moving internal wave fields. In both, a spectral description is used for the Bragg scatterers rather than a correlation description. This offers a simplified interpretation based on narrow-band processes. For the Kelvin wakes, a deterministic problem is solved in which the Bragg-scattering field exists (the diverging wake wave-field) in the presence of a long "corrugating" wave (the transverse wake wave-field). The Kelvin wake is assumed to be steady, and so the Bragg-scatterers are phase-locked to the corrugating wave: specifically, both have the same component of phase velocity along the ship's track, namely, the ship's speed. The corrugating field is not particularized to a detailed Kelvin wake structure but is left as an infinite sinusoid. The Bragg scatterers are also left unspecified in detail. The problem is solved in terms of a filter function which, when applied to the Bragg scattering divergent wave field, provides the unsquared complex SAR image. A numerical example is given for conditions pertinent to narrow-V wakes recorded in Dabob Bay, WN<sup>4</sup>.

made, and for the present purpose of discussing 1<sup>st</sup>-order Bragg backscatter, only terms linear in surface wave height will be kept. The ocean will also be treated as being a perfect conductor and only horizontal polarization in the incident radar field will be examined. These last two conditions can be removed but only with some additional algebraic complexity that does not illuminate the central issues of the present problems. The source and receiver will also be modelled as points, but with beam shaping introduced at a later stage.

With these approximations, the scattered field becomes

$$\vec{\psi}(\vec{x}, z, t) = \frac{z^2}{4\pi^2 c^2} \iint_{-\infty}^{\infty} \frac{\zeta(\vec{x}', t_R) \vec{S}''(t_S)}{r_1^2 r_2^2} d\vec{x}' \quad (1)$$

where  $\vec{\psi}$  is the vector radar field,  $z$  is the source-receiver altitude,  $\vec{x}(=x, y)$  is a coordinate frame located at and moving with the source-receiver,  $c$  is the speed of radar propagation in air,  $\vec{S}$  represents the radiated radar field (with units of  $\psi \cdot \text{length}$ ), and  $\vec{x}'(=x', y')$  is a non-moving coordinate frame fixed to the ambient ocean surface. Furthermore,

$$t_R = t - r_1/c \quad , \quad (2)$$

$$t_S = t_R - r_2/c - \frac{V}{c^2} (y' - vt) \approx t - \frac{r_1 + r_2}{c} \quad , \quad (3)$$

$$r_1 = \sqrt{(x - x')^2 + (y + Vt - y')^2 + z^2} \quad , \quad (4)$$

$$r_2 = \sqrt{x'^2 + (y' - Vt_R)^2 + z^2} \approx \sqrt{x'^2 + (y' - Vt)^2 + z^2} \quad , \quad (5)$$

$$\vec{S}''(t_S) = d^2 \vec{S}(t_S) / dt_S^2 \quad , \quad (6)$$

$$\zeta = \text{surface waveheight} \quad ,$$



and here  $V$  is the aircraft speed oriented along the  $y$ -direction (azimuthal), and the centre of the radar beam is oriented along the  $x$ -direction (range). The ratio  $V/c$  is sufficiently small that the approximate forms of Equations (3) and (5) will be used.

The  $r_1$ - and  $r_2$ -terms in the denominator of the integrand are sufficiently slowly varying and sufficiently similar that they can be removed from the integral and each set equal to  $R$ , the distance from the source-receiver to the centre of the scene element (i.e. pixel).  $\vec{S}$  is also narrowband enough in frequency that the approximation

$$\vec{S}'' = -\omega_c^2 \vec{S} \quad (7)$$

can be used with no appreciable loss of accuracy for the present purposes, with  $\omega_c$  being the central frequency in the radiated radar field.

The vector character of  $\vec{\psi}$  and  $\vec{S}$  appears only in the radiated fields and it is scalar versions of these that are used in the SAR processing. The conversions take place at the antenna and bring in the beam-shape function. The received voltage is given by

$$\begin{aligned} v(t) &= \iiint_{-\infty}^{\infty} \vec{F}_R(x, y, z) \cdot \vec{\psi}(x, y, z, t) dx dy dz \\ &= \iiint_{-\infty}^{\infty} \vec{F}_R(x, y, z) \\ &\quad \times \iint_{-\infty}^{\infty} \iint_{-\infty}^{\infty} \frac{z_0 z_s(\vec{x}', t_R) S''(t_s) \vec{F}_T(x_s, y_s, z_s)}{r_1^2(x, y, z, t, \vec{x}') r_2^2(x_s, y_s, z_0, t, \vec{x}')} d\vec{x}' dx_s dy_s dz_s dx dy dz \end{aligned} \quad (8)$$

Here,  $\vec{F}_T$  is the beam-shaping function for transmission,  $\vec{F}_R$  is the beam-shaping function for the receiver, and the vector character of the radiated field is included explicitly in  $\vec{F}_T$  with  $S_1$  now merely being the (scalar) time-dependent part. Also, the receiver altitude is now at  $z$  rather than being coincident with the transmitter at  $z_s$ . Equation (8) can be rewritten as

$$\phi(t) = \iiint_{-\infty}^{\infty} \vec{F}_R(x, y, z) \iiint_{-\infty}^{\infty} \vec{F}_T(x_s, y_s, z_s) \phi_p(x, y, z; x_s, y_s, z_s; t) dx dy dz dx_s dy_s dz_s \quad (9)$$

with

$$\phi_p(t) = \frac{zz_s}{4\pi^2 c^2} \iint_{-\infty}^{\infty} \frac{s(\vec{x}', t_R) S_1(t_s)}{r_1^2 r_2^2} d\vec{x}' \quad (10)$$

Because of the far-field condition for the scattering and all the preceding assumptions, Equation (9), with Equation (10) can be simplified into

$$\phi(t) = - \frac{z^2 \omega_0^2}{4\pi^2 c^2 R^4} \iint_{-\infty}^{\infty} s(\vec{x}', t_R) S(t_{s_0}; x', y', -vt, z) d\vec{x}' \quad (11)$$

where

$$S(t; x, y, z) = S_1(t) B(x, y, z) \quad (12)$$

and  $B$  is now the modulus squared of the voltage beam-shape factor. In this form, beam shape is incorporated directly into the scalar  $S$  which describes the incident radiation.

The SAR image is formed by matched-filtering  $\phi(t)$  with a version of  $\phi$  that would be obtained from a stationary point reflector. The matching is done over the range coordinate i.e., over a  $t$ -interval that corresponds to the radar field propagation time from the antenna to the most distant

scatterer to be imaged and back to the antenna. Matching is also done separately over azimuth and for that  $S$  and  $\phi$  are modelled as being pulse-trains and the matching summation is performed over a suitable number of contiguous pulses, a number which can represent an aircraft travel distance that is a significant fraction of the azimuthal beamwidth at the sea surface.

For an explicit pulse-train description, Equation (11) becomes

$$\phi_n(\tau) = - \frac{z^2 \omega_c^2}{4\pi^2 c^2 R^4} \int_{-\infty}^{\infty} s(\vec{x}, t_n + \tau - \frac{r_{1n}}{c}) S(\tau - \frac{r_{1n} + r_{2n}}{c}; x', y' - Vt_n, z_0) d\vec{x}' \quad (13)$$

where  $\tau$  is time measured from the start of the  $n^{\text{th}}$  pulse,  $t_n$ , and  $r_{1n}$  and  $r_{2n}$  are ranges at  $t_n$ . For most applications  $\tau$  can be dropped from the time-argument of  $s$  because the radar scattering interval is effectively only a small fraction of a millisecond, a time short enough that the surface height usually does not change appreciably. The same simplification has been made in the spatial argument of  $S$  (relating to the beam shape factor in azimuth), because the aircraft travels a very short distance in the radar propagation time and the beam shape is smooth enough that this distance is unimportant.

The point-reflector replica of  $\phi(t)$  is obtained by letting  $s$  be a fixed  $\delta$ -function at location  $(x_0, y_0)$ . The coordinate  $y_0$  is left unspecified for the present but it will eventually be set to zero because the processing will always be considered to be done symmetrically in azimuth about the location of the replica. At this stage also the aircraft speed mismatch can be included by using  $V_1$  rather than  $V$  in the replica description, giving

$$\phi_{\text{rep}}(\tau) = - \frac{z_0^2 \omega_c^2 \sigma_0}{4\pi^2 c^2 R_0^4} S(\tau - \frac{2\sqrt{x_0^2 + (y_0 - V_1 t_n)^2 + z_0^2}}{c}) \quad (14)$$

Here,  $\sigma_0$  is the scattering strength of the point reflector.

For a Gaussian beam shape and Gaussian pulse shape,  $S$  takes the form

$$S(t; x, y - Vt_n, z) \approx e^{-i\omega_c t - i\omega' t^2 - t^2/T_G^2 - (y - Vt_n)^2/Y_G^2} + \text{c.c.} \quad (15)$$

where  $T_G$  and  $Y_G$  are the Gaussian beamwidth factors in time and azimuth respectively and where c.c. indicates complex conjugate. Here also, frequency modulation is explicitly included by means of the  $\omega'$ -term, and the range coordinate beam-shape is ignored because it is usually very broad and contributes effectively only to a general shading in the overall image.

In carrying out the matching process,  $\phi_n(\tau)$  and  $\phi_{\text{rep}}(\tau)$  are both demodulated (frequency-shifted to base-band) and only one side of the transform is used (e.g. positive frequencies only). This results in some computational simplicity for digitally formed images although neither step is crucial to image formation. Appropriate range-dependent phase-shifting terms are also incorporated to eliminate extraneous modulation products in the result. The undetected image is given by

$$W(\vec{x}_0) = \sum_n \int_{-\infty}^{\infty} \phi_n(\tau) \phi_{\text{rep}}^*(\tau) d\tau \quad (16)$$

where the  $\phi$ 's are now modified as just described and the sum is over  $N$  contiguous pulses.

By using Equations (13) and (14), Equation (16) can be written

$$W(\vec{x}_0) \approx \sum_n \iint_{-\infty}^{\infty} s(\vec{x}', t_n - \frac{r_{1n}}{c}) \int_{-\infty}^{\infty} \left( -\frac{z^2 \omega_c^2}{4\pi^2 c^2 R^4} S_D(\tau - \frac{r_{1n} + r_{2n}}{c}; x', y' - Vt_n, z) \right. \\ \left. S_D^*(\tau - \frac{2\sqrt{x_0^2 + (y_0 - Vt_n)^2 + z_0^2}}{c}; x', y' - Vt_n, z_0) \right) e^{\frac{2i\omega_c x_0^2}{c\sqrt{x_0^2 + z_0^2}}} dx' dy' \quad (17)$$

where the subscript D on each S indicates demodulation, and positive frequencies only, and the unimportant amplitude scale factor term in  $\phi_{rep}$ , i.e.  $z_o^2 \omega_c^2 \sigma_o / 4\pi^2 c^2 R_o^4$ , has been dropped. The purely phase-shifting term  $\exp(2i\omega_c x_o^2 / c \sqrt{x_o^2 + z_o^2})$  is also shown explicitly (although it need not be inside the integral).

For analytical simplicity, the sum on n is approximated as an integral over  $t_n$  using the definition

$$t_n = n\Delta \quad (18)$$

where  $\Delta$  is the time interval between successive radiated pulses. With this transformation, the sum becomes

$$\sum_n \rightarrow \frac{1}{\Delta} \int \dots dt_n \quad (19)$$

One further simplifying change can be made, shifting from  $t_n$  to the time-argument,  $t'$ , of  $s$ , i.e.

$$t' = t_n - r_{1n}/c \quad (20)$$

To the desired degree of accuracy, and using an impulse response formulation, Equation (17) becomes

$$W(\vec{x}_o) = \iiint_{-\infty}^{\infty} s(\vec{x}', t') I(\vec{x}', \vec{x}_o, t') d\vec{x}' dt' \quad (21)$$

where

$$I(\vec{x}, \vec{x}_0, t') = \frac{-z^2 \omega_c^2}{4\pi^2 c^2 R^4 \Delta} \exp \left\{ \frac{2i\omega_c}{c} \frac{x_0^2}{\sqrt{x_0^2 + z_0^2}} \right\} \int_{-\infty}^{\infty} S_D \left( \tau - \frac{r_1 + r_2}{c}; x', y' - Vt' - \frac{Vr_1}{c}, z \right) \\ \times S_D^* \left( \tau - \frac{2\sqrt{x_0^2 + (y_0 - V_1 t' - V_1 r_1/c)^2 + z_0^2}}{c}; x', y' - V_1 t' - \frac{V_1 r_1}{c}, z_0 \right) d\tau \quad (22)$$

and

$$S_D(t - \epsilon; x, y, z) = e^{\frac{i\epsilon\omega_c}{c} - i\omega'(t - \epsilon)^2 - \frac{(t - \epsilon)^2}{T_G^2} - \frac{y^2}{Y_G^2}} \quad (23)$$

The  $r$ 's in Equations (20) and (22) are actually functions of  $t$  (see Equations (4) and (5)) but to the required accuracy they may be treated as time-independent, particularly as used in Equation (22). This can be shown by expanding  $t_R$  and  $t$  about  $t_n$  and substituting these expansions into the expressions for  $r_1$  and  $r_2$  and ignoring terms of order  $V/c$  that are small.

With all the approximations incorporated, the final expression for  $I$  becomes

$$I(\vec{x}, \vec{x}_0, t') = - \frac{z^2 \omega_c^2}{4\pi^2 c^2 R^4 \Delta} \exp \left[ \frac{2i\omega_c}{c} \frac{x_0^2}{\sqrt{x_0^2 + z_0^2}} + \frac{i\omega_c}{c} (r_1 + r_2 - 2r_0) \right. \\ \left. - \left\{ \frac{(y' - V_1 t')^2 + (y' - Vt')^2}{Y_G^2} \right\} - \frac{(r_1 + r_2 - 2r_0)^2}{2c^2 T_G^2} (\omega'^2 T_G^4 - 1) \right] \quad (24)$$

where

$$r_1 \approx \sqrt{x'^2 + (y' - Vt')^2 + z^2} \quad , \quad (25)$$

$$r_2 \approx r_1 \quad , \quad (26)$$

and

$$r_0 = \sqrt{x_0^2 + (y_0 - v_1 t')^2 + z_0^2} \quad (27)$$

This completes the general analysis. Fourier transform descriptions for  $\varsigma$  will be used in the following sections but with forms that are specifically relevant for each problem. Also, the  $r$ 's of Equations (25) to (27) will be binomially expanded about a common position and terms up to and including second order in deviations from this position will be kept. For the Kelvin wake problem,  $z$  in Equation (25) will also be allowed to be a function of position in order to describe the effect of the height of the corrugating wave.

### 3. Kelvin Wakes

The entire wake wave-field including the corrugating portion is reducible to a steady-state by a simple transformation to coordinate axes moving with the source ship. Thus, with  $\vec{c}_s$  as the ship's velocity, and with distance measured from the ship being given by  $\vec{\eta}$ , the following pertain:

$$\vec{\eta} = \vec{x}' - \vec{c}_s t' \quad , \quad (28)$$

$$\varsigma(\vec{x}', t') = \varsigma(\vec{\eta}) \quad ,$$

and, from Equation (21)

$$W(\vec{x}_0) = \iint_{-\infty}^{\infty} \varsigma(\vec{\eta}) \int_{-\infty}^{\infty} I(\vec{\eta} + \vec{c}_s t', \vec{x}_0, t') dt' d\vec{\eta} \quad (29)$$

The divergent wave-field  $\varsigma$  can now be expressed as a two-dimensional Fourier transform,

$$\varsigma(\vec{\eta}) = \iint_{-\infty}^{\infty} p(\vec{k}) e^{i\vec{k} \cdot \vec{\eta}} d\vec{k} \quad (30)$$

so that

$$p(\vec{k}) = \frac{1}{4\pi^2} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \varsigma(\vec{\eta}) e^{-i\vec{k} \cdot \vec{\eta}} d\vec{\eta} \quad , \quad (31)$$

$$\text{and } W(\vec{x}_0) = \int_{-\infty}^{\infty} p(\vec{k}) e^{i\vec{k} \cdot \vec{x}_0} \left( \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} e^{i\vec{k} \cdot (\vec{\eta} - \vec{x}_0)} I(\vec{\eta} + \vec{c}_s t', \vec{x}_0, t') dt' d\vec{\eta} \right) d\vec{k} \quad (32)$$

The term  $e^{i\vec{k} \cdot \vec{x}_0}$  is taken out explicitly in the integrand of this last expression in order that  $W(\vec{x}_0)$  may be seen to be simply a filtered version of  $\varsigma$  by comparing Equations (32) and (30). The filter function, denoted  $Q_2$ , is given by the term in the braces in Equation (32), i.e.,

$$Q_2(\vec{k}; \vec{x}_0) = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} e^{i\vec{k} \cdot (\vec{\eta} - \vec{x}_0)} I(\vec{\eta} + \vec{c}_s t', \vec{x}_0, t') dt' d\vec{\eta} \quad (33)$$

and it contains all of the frequency-shaping, phase-shifting and demodulation that the SAR process applies to the  $\varsigma$ -field in the Fourier domain. In this steady-state formulation, all of the fluid-dynamically interactive effects of the corrugating field and its currents on the  $\varsigma$ -field are contained in  $p(\vec{k})$  and need not be expressed explicitly. This simple situation will not hold in the next section which deals with internal wave modulations that have a stochastic nature and there the frequency-shifting effect of the currents in the corrugating field will need to be explicitly incorporated.

To progress further with the Kelvin wake problem it is necessary to return to Equations (24) to (27) and expand the expressions for  $r_1, r_2$  and  $r_0$  using the definitions

$$x' = x_0 + \mu \quad , \quad (34)$$

$$y' = Vt' + y'' \quad , \quad (35)$$

$$y_0 = V_1 t' + q'' \quad , \quad (36)$$

$$z = z_0 + \varsigma_T(\vec{\eta}) \quad , \quad (37)$$

$$R_0 = \sqrt{x_0^2 + z_0^2} \quad . \quad (38)$$



With terms expressed up to and including second order in  $\mu$ ,  $y''$ ,  $q''$  and  $s_T$ ,

$$r_1 - r_2 = R_0 + \frac{\mu x_0 + s_T z_0}{R_0} + \frac{u^2 + y''^2 + s_T^2}{2R_0} - \frac{x_0^2 \mu^2}{2R_0^3} - \frac{\mu s_T x_0 z_0}{R_0^3} - \frac{z_0^2 s_T^2}{2R_0^3} + \dots \quad (39)$$

$$r_0 = R_0 + \frac{q''^2}{2R_0} + \dots \quad (40)$$

and, from Equations (24) and (34) to (36)

$$I(\vec{x}', x_0, t') = I(\mu, y'', \vec{x}_0, q'') \quad (41)$$

$$= - \frac{z_0^2 \omega_c^2}{4\pi^2 \omega_c^2 R_0^4 \Delta} \exp \left[ \frac{2i\omega_c}{c} \frac{x_0^2}{R_0} + \frac{i\omega_c}{c} \Phi(\mu, y'', q'') - \frac{y''^2 + (y'' + (V-V_1)/V_1 (y_0 - q''))^2}{Y_G^2} - (\mu x_0 + s_T z_0)^2 (\omega'^2 T_G^4 - 1) / 2c^2 T_G^2 R_0^2 \right] \quad (42)$$

where

$$\Phi(\mu, y'', q'') = r_1 + r_2 - 2r_0 = \frac{2\mu x_0 + s_T z_0}{R_0} + \frac{\mu^2 z_0^2}{R_0^3} + \frac{y''^2 - q''^2}{R_0} + \frac{s_T^2 x_0^2}{R_0^3} - \frac{2\mu s_T x_0 z_0}{R_0^3} \quad (43)$$

and where, by Equations (28), (34) and (36)

$$\mu = \eta_x - x_0 + c_{sx}(y_0 - q'')/V_1, \quad (44)$$

$$y'' = \eta_y + (c_{sy} - V)(y_0 - q'')/V_1. \quad (45)$$

Here,  $\eta_x$  and  $\eta_y$  are the components of  $\vec{\eta}$ , and  $c_{sx}$  and  $c_{sy}$  are components of  $\vec{c}_s$ .

The integration in  $t'$  in Equation (33) can be performed after converting  $t'$  to  $q''$  with the use of Equation (36), and the result is

$$Q_2(\vec{k}; \vec{x}_0) = \iint_{-\infty}^{\infty} e^{ik\vec{\eta}(\eta-x_0)} K(\vec{\eta}, \vec{x}_0) d\vec{\eta} \quad (46)$$

where

$$K(\vec{\eta}, \vec{x}_0) = - \frac{z_0^2 \omega_c^2}{4\pi^2 c^2 R_0^4 \Delta V_1} \frac{\pi T_G}{\sqrt{2|C_1|}} e^{A_1 + B_1^2/4C_1} \quad (47)$$

To obtain this expression, the argument of the exponential in Equation (42) has been written as

$$\arg = A_1 + B_1 q'' - C_1 q''^2 \quad (48)$$

and, by substitution of Equations (44) and (45) into (43) and (42) and subsequent inspection, the coefficients  $A_1$ ,  $B_1$  and  $C_1$  can be readily obtained. They will not be expressed explicitly here, but, because of the original quadratic-form expansions of  $r_1$ ,  $r_2$  and  $r_0$ , it can be noted that  $\arg$  itself is at most quadratic in products of  $\vec{\eta}$  and  $q''$ , and so,  $A_1$  is quadratic in  $\vec{\eta}$ ,  $B_1$  is linear in  $\vec{\eta}$  and  $C_1$  is constant. The same orders exist for  $s_T$ . A further simplification will also be made to these coefficients, namely that terms quadratic in the small quantities  $|\vec{c}_s|/V$  and  $s_T/R_0$  will be ignored. For the usual range of SAR parameters, these quantities are of order 0.1 or less and the ignored quadratic terms are always dominated by linear counterparts.

Without specifying the functional form of  $s_T$  on  $\vec{\eta}$ , the Fourier transform represented by Equation (46) cannot be carried out. However, insight can be gained into possible effects of  $s_T$  in the resulting image, by examining the special case in which

$$c_{sx} = 0 \quad (49)$$

i.e., the source ship travelling parallel to the SAR aircraft track, and

$$\varsigma_T = \varsigma_0 + \eta_y \left( \frac{\partial \varsigma_T}{\partial y} \right)_0 + \frac{1}{2} \eta_y^2 \left( \frac{\partial^2 \varsigma_T}{\partial y^2} \right)_0 \quad (50)$$

(For simplicity, the processor mismatch is also set equal to zero, i.e.,  $V=V_1$ .)

The surface underlying the Bragg-wave is thus given a tilt and a curvature in the azimuthal direction only. The argument of the exponent in Equation (47) can be put in a quadratic normal form

$$\arg = A - [(\eta_x - x_s)/\text{Res}_x]^2 \ln 6 - [(\eta_y - y_s)/\text{Res}_y]^2 \ln 6 \quad (51)$$

where  $\text{Res}_x$  and  $\text{Res}_y$  represent full-widths at half-maximum and  $x_s$  and  $y_s$  represent the position in  $\vec{\eta}$  at which the maximum occurs. For the case presented by Equations (49) and (50),

$$\text{Res}_x = \frac{\frac{cR_0}{\omega' x_0 T_G} \sqrt{\ln 4}}{(1 + 1/\omega'^2 T_G^4)^{1/2}} \quad (52)$$

$$\text{Res}_y = \frac{\frac{2cR_0}{\omega_c T_G} \sqrt{\ln 4}}{(1 - 4 \left( \frac{\omega' T_G}{\omega_c} \right)^2 \left( \frac{x_0}{Y_G} \right)^2 z_0 \left( \frac{\partial^2 \varsigma_T}{\partial y^2} \right)_0)^{1/2}} \left( 1 + \frac{c s y}{V} \right) \quad (53)$$

The expression for  $\text{Res}_x$  is unmodified by the presence of  $\varsigma_T$  (to this order), but it can be seen that  $\text{Res}_y$  can be significantly altered by  $\varsigma_T$ 's curvature. This gives rise to the possibility of brightened spots appearing in a wake image at positions where there is a coincidence of maximum (positive)  $\partial^2 \varsigma_T / \partial y^2$ , and Bragg scatterers in the diverging wake components. For stronger conclusions the form of  $\varsigma_T$  will be specified more fully and numerical evaluations will be performed.

The specific values chosen for numerical treatment pertain to the Canada Centre for Remote Sensing L-band system (originally constructed by the Environmental Research Institute of Michigan) as configured in July 1983 and used in overflights<sup>4</sup> of Dabob Bay, WN. For JOWIP 8/2

$$\omega_c = 2\pi (1.185)10^9 \text{ rad/s} \quad (54a)$$

$$\omega' = 2\pi (33.3)10^{12} \text{ rad/s} \quad (54b)$$

$$x_o = 3033.5 \text{ m} \quad (54c)$$

$$z_o = 7010.0 \text{ m} \quad (54d)$$

$$y_o = 0 \text{ m} \quad (54e)$$

$$Y_G = 162.1 \text{ m} \quad (54f)$$

$$T_G = 0.951 \cdot 10^{-6} \text{ s} \quad (54g)$$

where the last two values are Gaussian equivalents which result in unperturbed resolution widths of  $\sqrt{20}$  m in both directions. For JOWIP 8/2,

$$V = 134 \text{ m/s} \quad (54h)$$

$$c_s = 8 \text{ m/s along a direction } \sim 13^\circ \text{ off azimuth} \quad (54j)$$

and the amplitude of the corrugating wave had a value of

$$\zeta_o \approx 0.1 \text{ m} \quad (54k)$$

at a distance of  $\sim 2.5$  km behind<sup>9</sup> the source ship (USS QUAPAW). For this ship speed, the wavelength of the corrugating transverse wake component is

$$\lambda_s = 41 \text{ m} \quad (54m)$$

calculated from the gravity wave formula  $2\pi c_s^2/g$ .

The corrugating wavefield  $\varsigma_T$  is modelled as a simple sinusoid. This idealization is used because variations in actual  $\varsigma_T$ 's perpendicular to the wake axis are minimal, particularly near the axis where the L-band Bragg waves are located in the diverging field. Thus,

$$\varsigma_T = \varsigma_0 \cos\left[\frac{2\pi}{\lambda_s}(\eta_x \sin\theta + \eta_y \cos\theta) + \phi\right] \quad (55)$$

where  $\theta$  is the angle of the ship's track with respect to the aircraft track ( $13^\circ$  for the present example), and  $\phi$  is a parameter that allows the full range of phase of  $\varsigma_T$  to be investigated numerically.

In the Fourier definition of  $Q_2$ , Equation (46), the  $\vec{\eta}$ -coordinate system can be rotated through the angle  $\theta$  so that  $\varsigma_T$  becomes a function of only one variable. Integration can be performed along the orthogonal component of the rotated  $\vec{\eta}$  system, and since the exponential argument of  $K$  is a quadratic form in  $\vec{\eta}$ , apart from  $\varsigma_T$ , it is also a quadratic form in the rotated system, and the integration simply returns a counterpart quadratic form. The remaining integral, along the direction  $\theta$  must be done numerically.

In the absence of a corrugating perturbation,  $Q_2$  is a general two-dimensional Gaussian form in  $\vec{k}$  and it can be described by its maximum height, the wavenumber position at maximum, and the widths along the  $\vec{k}$ -axes to the half-maximum points. This is illustrated in contour form in Figure 1. If the peak is also rotated, the angle of rotation is necessary to complete the description. In the presence of the perturbation, these parameters change, and the effect of the perturbation can be described fully by the amounts of the changes. Figure 2 shows the changes in form of the half-maximum contour of  $Q_2$  for different phases, i.e. different values of  $\phi$  in Equation (55), along the corrugating wave. Figures 1 and 2 both use the parameters specified in Equations (54a) to (54m) except for Equation (54k). Figure 1 uses  $\varsigma_0=0$  and Figure 2 uses  $\varsigma_0=0.5$  m.

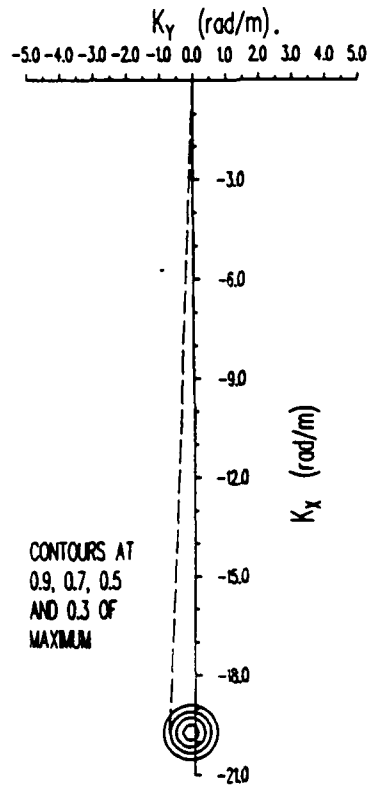


Figure 1. Contours of  $Q_2$  in the absence of a corrugating wave (i.e.  $\zeta_0=0$ ) but with the source travelling along  $\theta=13^\circ$ . The 1st-Bragg wavenumber for this case is  $(-19.712\dots, -0.13238\dots)$  rad/m. The dotted line indicates the extreme in propagation direction ( $2.20^\circ$ ) for  $Q_2$  at its half-maximum.

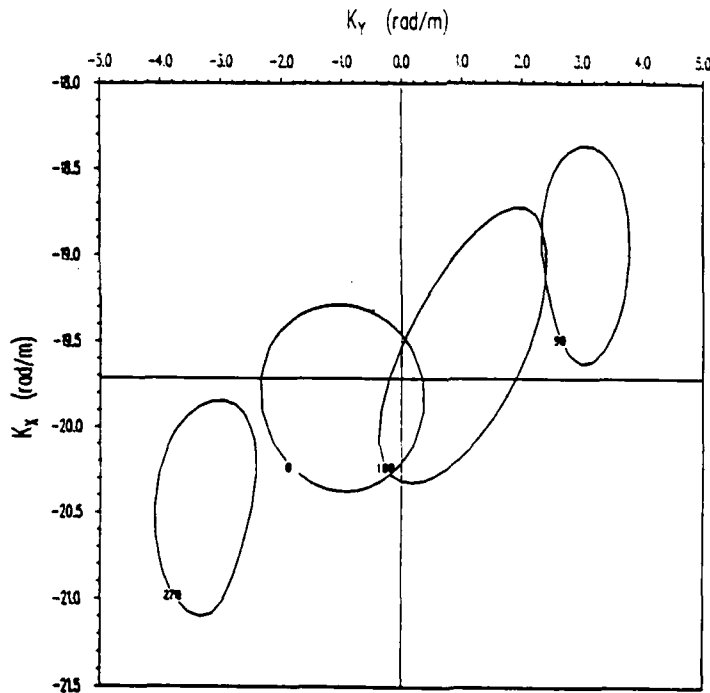


Figure 2. The half-maximum contour of  $Q_2$  for various phases (degrees) along a corrugating wave of amplitude  $\zeta_0=0.5$  m propagating at  $\theta=13^\circ$  to the azimuthal direction. Note different scales from Figure 1.

Without a corrugating wave, it can be seen from Figure 1 that surface waves with a wavenumber centered at  $-19.712\dots, -0.132$  rad/m and within a circular band of width  $\pm 0.6240\dots$  rad/m are accepted by the processor with a gain of between 1 and 0.5. This bandwidth implies that all waves propagating at an angle within  $\pm 1.81^\circ$  centered at  $0.38^\circ$  off this range direction and with a  $k_x$ -value of  $-19.712$  rad/m, are accepted by the processor with a gain reduction of no more than a factor of 2. (The slight offset of the peak from  $k_y=0$  is due to the non-zero value of  $\theta$ , i.e. the non-zero value of  $c_{sx}$ . This range-component of motion in the Bragg scattering wavefield coupled with the aircraft motion and time interval between successive radar pulses results in a slight skewing of the Bragg-wave field in the SAR pattern.)

With a corrugating wave as used in Figure 2, it can be seen that the surface wave wavenumbers accepted by the processor vary over a much more considerable range, allowing waves propagating at up to  $11.5^\circ$  to the range direction to be "seen" (at  $\phi=90^\circ$ ), with a gain reduction of no more than a factor of 2 from the peak value. This variation is depicted more fully in Figure 3 where the extreme  $k_y$ -value for the half-maximum contour is shown as a function of  $\phi$  for  $s_0$ -values from 0 to 0.5 m. In a similar fashion, extreme  $k_x$ -values are shown in Figure 4, and the  $Q_2$ -value at its peak is shown in Figure 5, for parameters as in Figure 3.

From Figure 3 it can be seen that the extreme  $k_y$ -values are largest for  $\phi=110^\circ$  (and smallest at  $290^\circ$ ). Using the central  $k_x$ -value of  $-19.712$  rad/m, the extreme values represent surface waves propagating at angles with respect to the range direction as shown in Figure 6. Here the propagation angles are given as a function of  $s_0$ , and propagation directions that pertain to the center of the processor peak are also given. The displayed data shows strongly linear dependence with  $s_0$ , and very little variation for  $\theta$ -values over the range  $\theta=0^\circ$  to  $13^\circ$ .

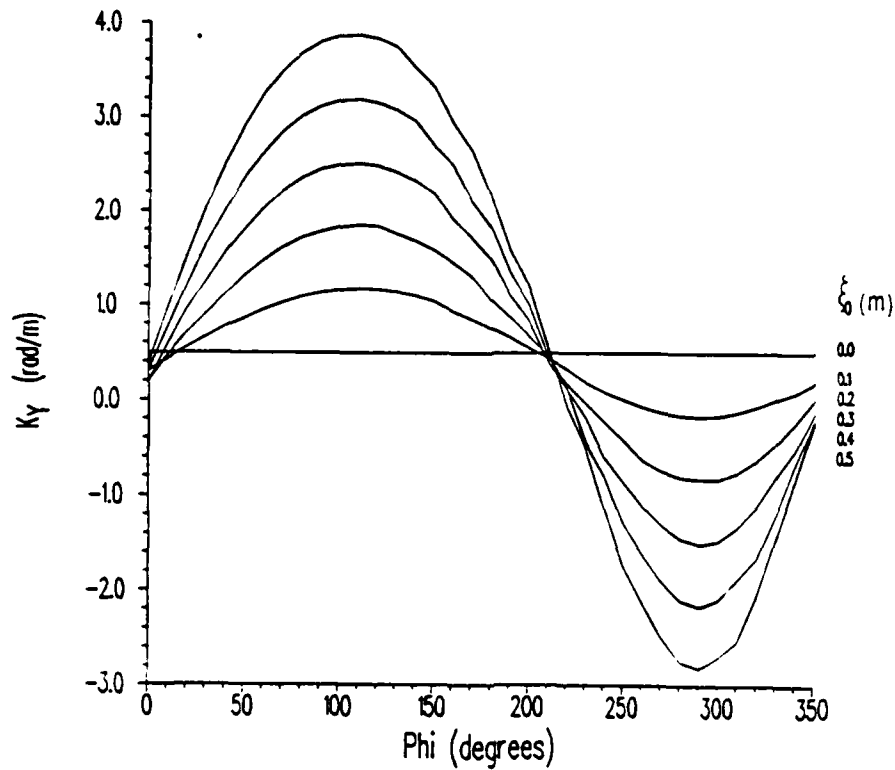


Figure 3. Variations of azimuth wavenumber location for the half-maximum  $Q_2$ -contour. Parameter values as given in Equation (54).

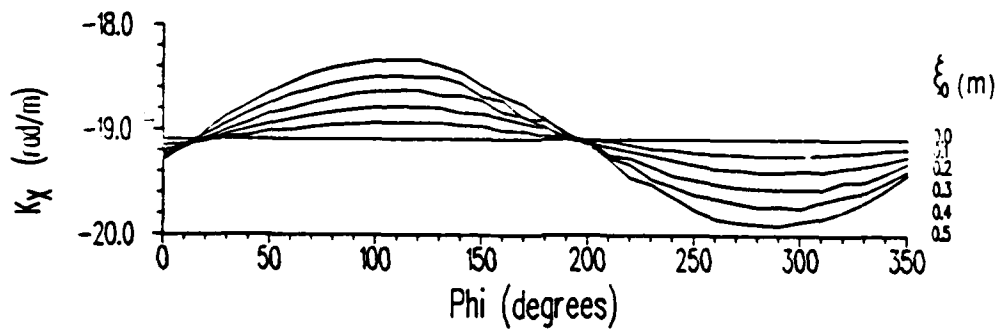


Figure 4. Variations of range wavenumber location for the half-maximum  $Q_2$ -contour for the case of Figure 3.

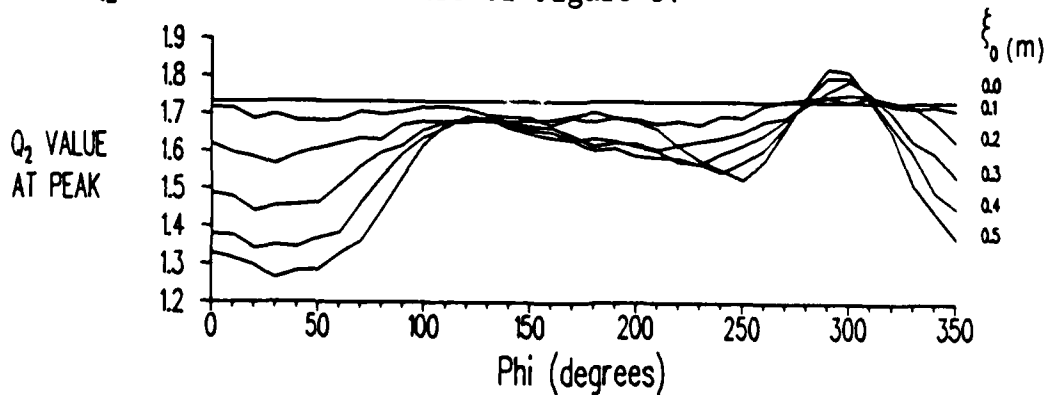


Figure 5. Variation of peak  $Q_2$ -value for the case of Figure 3.



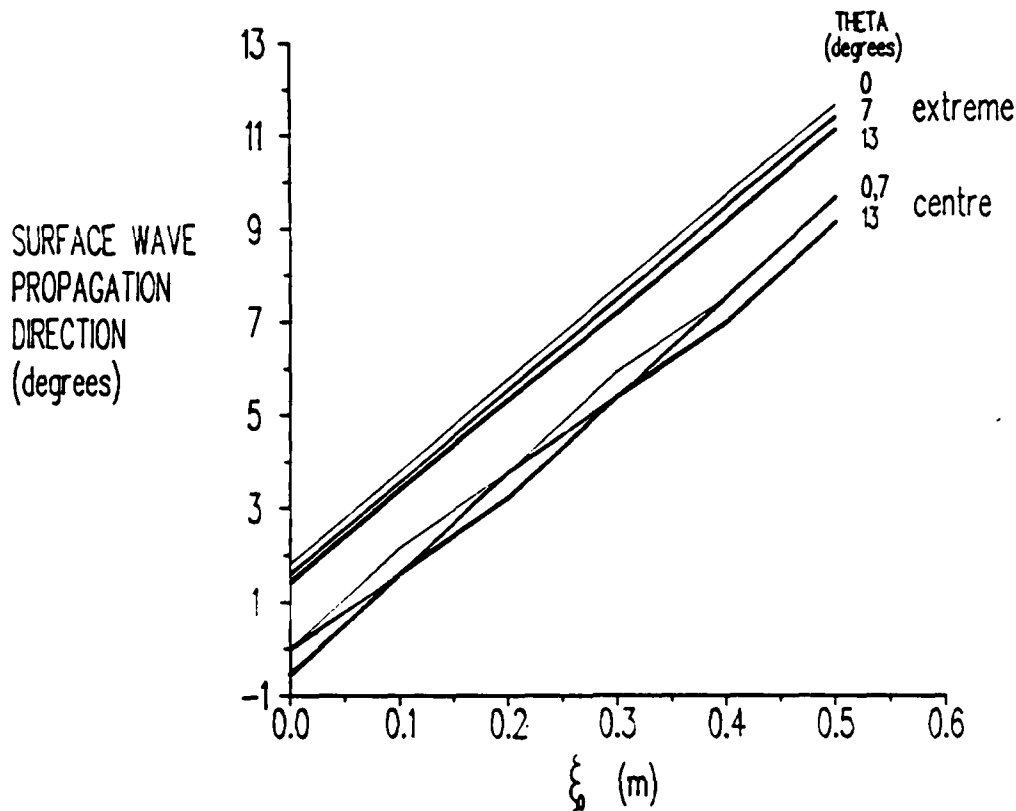


Figure 6. Surface wave propagation direction (as measured from the range coordinate) versus amplitude  $\xi_0$  of the corrugating wave.

The peak values of  $Q_2$  as given in Figure 5 vary only over the range 3.35 to 4.90 for  $\xi_0=0.5$  metres, only 1.6 dB, and so the direct variation of  $Q_2$  is a relatively minor amount. For the Kelvin wake problem, the periodic variation of the filter window in  $k_y$ -space is potentially much more significant.

For experimental comparisons, there are three images of clear narrow-V wakes that were taken during the JOWIP measurements, those designated JOWIP 8/2, 8/3 and 9/2. They are shown in Figure 7. From these images the orientation of the most steeply inclined narrow-V arms for each pass have been measured, with respect to the SAR aircraft track, and the values are given in Table 1. Ship track angles as determined from the positioning equipment<sup>10</sup> are also included for each of these runs.

Table 1. Ship Track and Steepest Wake Arm Inclination to SAR Track

SAR Pass	Wake Arm Angle	Ship Track Angle
8/2	18°	13°
8/3	15°	11°
9/2	11°	9°

It can be seen from the Table that the wake arm inclinations are of the same order or are larger than the processor acceptance angles (at half-maximum) for  $\zeta_0=0.5$  m. However, the variational effects just described will contribute strongly toward enhanced (or decreased) outputs in the SAR image even for these cases.

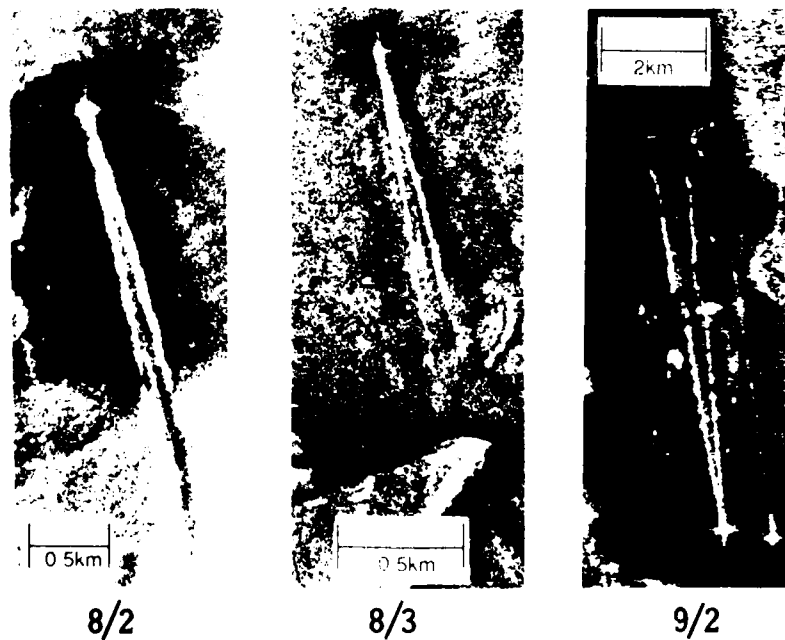


Figure 7. SAR images of narrow-V wakes

#### 4. Internal Waves

In order to determine the effective impulse response function of the SAR system as it pertains to propagating internal wave modulations of surface wave fields, a Fourier description for  $\varsigma$  will be used in Equation (21) with modulational features explicitly incorporated. The surface wave field will be specified using a stochastic description rather than a deterministic one as used in the last section. Let  $\varsigma$  be described in terms of linear dispersion theory, i.e. frequency  $\sigma$  being a given function of wavenumber  $\vec{k}$ , but with modulating transform terms A:

$$\begin{aligned} \varsigma(\vec{x}', t') = & \int_{-\infty}^{\infty} \int \int p_s(\vec{k}) A(\vec{k}, \sigma_f; \vec{x}', t') \delta(\sigma - \sigma_f - \vec{k} \cdot \vec{U}(\vec{x}', t')) e^{i\vec{k} \cdot \vec{x}' - i\sigma t'} d\vec{k} d\sigma \\ & + \int_{-\infty}^{\infty} \int \int p_s^*(-\vec{k}) A^*(-\vec{k}, \sigma_f; \vec{x}', t') \delta(\sigma + \sigma_f - \vec{k} \cdot \vec{U}(\vec{x}', t')) e^{i\vec{k} \cdot \vec{x}' - i\sigma t'} d\vec{k} d\sigma \end{aligned} \quad (56)$$

Here  $p_s(\vec{k})$  is a wide sense stationary transform,  $A(\vec{k}, \sigma_f; \vec{x}', t')$  describes the effect of the internal wave modulation on the surface wave field,  $\vec{U}$  is the horizontal surface current (assumed to be due to the internal wave field only), and  $\sigma_f$  is the radian frequency of freely propagating surface waves, i.e.

$$\sigma_f = \sqrt{gk + Tk^3} \quad . \quad (57)$$

In this last equation,  $g$  is the acceleration of gravity and  $T$  is the surface tension of the air-water interface. In Equation (56),  $\varsigma$  is guaranteed to be real by the form of the right-hand side, and the surface wave propagation properties in the presence of the advecting current field  $\vec{U}$  are accounted for by the  $\delta$ -functions.

Equation (56) can be substituted into Equation (21) with the result

$$W(\vec{x}_0) = \iint_{-\infty}^{\infty} [p_s(\vec{k}) \{ \iiint_{-\infty}^{\infty} A(\vec{k}, \sigma_f; \vec{x}', t') I(\vec{x}', \vec{x}_0, t') e^{i\vec{k} \cdot \vec{x}' - i\sigma_f t' - i\vec{k} \cdot \vec{U}' t'} d\vec{x}' dt' \} + p_s^*(-\vec{k}) \{ \iiint_{-\infty}^{\infty} A^*(-\vec{k}, \sigma_f; \vec{x}', t') I(\vec{x}', \vec{x}_0, t') e^{i\vec{k} \cdot \vec{x}' + i\sigma_f t' - i\vec{k} \cdot \vec{U}' t'} d\vec{x}' dt' \} ] d\vec{k} \quad (58)$$

and here,  $\vec{U}'$  is an abbreviated notation for  $\vec{U}(\vec{x}', t')$ .

The image  $I(\vec{x}_0)$  is defined as

$$I(\vec{x}_0) = |W(\vec{x}_0)|^2 \quad (59)$$

and so the mean image is given by

$$\langle I(\vec{x}_0) \rangle = \langle |W(\vec{x}_0)|^2 \rangle, \quad (60)$$

i.e.,

$$\begin{aligned} \langle I(\vec{x}_0) \rangle = & \iint_{-\infty}^{\infty} P_s(\vec{k}) \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} A(\vec{k}, \sigma_f; \vec{x}', t') A^*(\vec{k}, \sigma_f; \vec{x}'', t'') I(\vec{x}', \vec{x}_0, t') I^*(\vec{x}'', \vec{x}_0, t'') \\ & + \iint_{-\infty}^{\infty} P_s^*(-\vec{k}) \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} A(-\vec{k}, \sigma_f; \vec{x}'', t'') A^*(-\vec{k}, \sigma_f; \vec{x}', t') I(\vec{x}', \vec{x}_0, t') \\ & \times I^*(\vec{x}'', \vec{x}_0, t'') e^{i\vec{k} \cdot (\vec{x}' - \vec{x}'') + i\sigma_f(t' - t'') - i\vec{k} \cdot (\vec{U}' t' - \vec{U}'' t'')} d\vec{x}' dt' d\vec{x}'' dt'' d\vec{k} \quad (61) \end{aligned}$$

where  $P_s(\vec{k})$  is the power spectrum of the  $p_s(\vec{k})$  process, defined by

$$\langle p_s(\vec{k}) p_s^*(\vec{k}') \rangle = P_s(\vec{k}) \delta(\vec{k} - \vec{k}') \quad (62)$$

In order to proceed further, it is helpful to express A by a Fourier transform and I by a Fourier-like transform:

$$A(\vec{k}, \sigma_f; \vec{x}', t') = \iiint_{-\infty}^{\infty} e^{i\vec{K} \cdot \vec{x}' - i\Omega t'} \phi_B(\vec{K}, \Omega; \vec{k}, \sigma_f) d\vec{K} d\Omega \quad (63)$$

$$Q_2(\vec{K}', \Omega', \vec{K}''; \vec{x}_0) = \iiint_{-\infty}^{\infty} I(\vec{x}', \vec{x}_0, t') e^{i\vec{K}' \cdot \vec{x}' - i\Omega' t' - i\vec{K}'' \cdot \vec{U}' t'} d\vec{x}' dt' \quad (64)$$

With these substitutions Equation (61) becomes,

$$\begin{aligned} \langle I(\vec{x}_0) \rangle = & \iiint_{-\infty}^{\infty} P_s(\vec{k}) \left| \iiint_{-\infty}^{\infty} \phi_B(\vec{K}, \Omega; \vec{k}, \sigma_f) Q_2(\vec{k} + \vec{K}, \sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega \right|^2 d\vec{k} \\ & + \iiint_{-\infty}^{\infty} P_s(-\vec{k}) \left| \iiint_{-\infty}^{\infty} \phi_B^*(\vec{K}, \Omega; -\vec{k}, \sigma_f) Q_2(\vec{k} - \vec{K}, -\sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega \right|^2 d\vec{k} \end{aligned} \quad (65)$$

and at this stage it is expedient to specialize to weak interactions so that

$$A(\vec{k}, \sigma_f; \vec{x}', t') = 1 + A_1(\vec{k}, \sigma_f; \vec{x}', t') \quad ; \quad A_1 \ll 1 \quad (66a, b)$$

Therefore,

$$\phi_B = \delta(\vec{K}) \delta(\Omega) + \phi_{A1}(\vec{K}, \Omega; \vec{k}, \sigma_f) \quad (67)$$

and the squared-term in the first integrand in Equation (65) becomes

$$\left| Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0) + \iiint_{-\infty}^{\infty} \phi_{A1}(\vec{K}, \Omega; \vec{k}, \sigma_f) Q_2(\vec{k} + \vec{K}, \sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega \right|^2$$

Because of Equation (66b) this becomes

$$|Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0)|^2 + Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0) \int_{-\infty}^{\infty} \int \phi_{A1}^*(\vec{K}, \Omega; \vec{k}, \sigma_f) Q_2^*(\vec{k} + \vec{K}, \sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega$$

$$+ Q_2^*(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0) \int_{-\infty}^{\infty} \int \phi_{A1}(\vec{K}, \Omega; \vec{k}, \sigma_f) Q_2(\vec{k} + \vec{K}, \sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega$$

and the remaining terms are ignorable, being  $O(A_1^2)$  cf 1. The squared term in the second integrand transforms in a similar manner. With this substituted in Equation (65), the term in  $|Q_2|^2$  can be seen to be simply  $\langle I(\vec{x}_0) \rangle_s$ , i.e. the image that would have existed in the absence of any modulation, and so, the deviation, due only to the modulational effects, is given by

$$\langle I(\vec{x}_0) \rangle - \langle I(\vec{x}_0) \rangle_s = \int_{-\infty}^{\infty} \int P_s(\vec{k}) (Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0)$$

$$\times \int_{-\infty}^{\infty} \int \phi_{A1}^*(\vec{K}, \Omega, \vec{k}, \sigma_f) Q_2^*(\vec{k} + \vec{K}, \sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega) d\vec{k}$$

$$+ \int_{-\infty}^{\infty} \int P_s(-\vec{k}) (Q_2(\vec{k}, -\sigma_f, \vec{k}; \vec{x}_0)$$

$$\times \int_{-\infty}^{\infty} \int \phi_{A1}(\vec{K}, \Omega, -\vec{k}, \sigma_f) Q_2^*(\vec{k} + \vec{K}, -\sigma_f + \Omega, \vec{k}; \vec{x}_0) d\vec{K} d\Omega) d\vec{k}$$

$$+ \text{c.c.} \quad (68)$$

The term  $Q_2$  defined in Equation (64) differs only from the corresponding term of Equation (33) by the presence of the phase-shifting component  $K \cdot \vec{U} t'$  due to the internal wave current. However for typical current values, i.e. 1 m/s, and even for large processor integration times, i.e. 10 seconds, the product  $U t'$  is equivalent to only a 10-metre shift in position. This is to be compared with a pixel size of 1 to 10 m in range (the range compression is already included in  $I$  in Equation (64)) or a

beamwidth of order 100 m in azimuth. For Gaussian beams,  $I$  is a Gaussian shape, and so with the  $K'' \vec{U}' t'$  term included up to second order in the  $\vec{x}', t'$  variables (see equivalent analysis in Section 3, Equations (39) to (43)),  $Q_2$  can be determined explicitly and it also is a Gaussian function in  $K$ -space. Because  $Q_2$  is essentially a Fourier transform of a Gaussian function with  $\sim 10$ -m widths (after the  $t'$ -integration), it displays widths in  $K$ -space of order  $2/10$  rad/m and these are centered at its Bragg wavenumber,  $\vec{k}_B$ . Again, using the numerical values of Equations (54a-g),  $\vec{k}_B$  is 19.7 rad/m along the range direction and zero along azimuth, and so, the bandwidth of  $Q_2$  in  $k$ -space (as used in Equation (68)) is only  $\sim 1\%$  of its central value, i.e. extremely narrow.

From the interaction dynamics, it can be determined that  $\phi_{A1}$  is a broad function of  $\vec{k}$ , and from estimates of surface wave spectra it is known that  $P_s(\vec{k})$  is broad and smooth over bandwidths of order  $1\%$ . These functions can then be approximated by their values at  $\vec{k}_B$  and removed from the  $k$ -integration processes in Equation (68) with very little error. For further analysis the integrands are returned to  $A_1^*$  and  $I$  using Equations (64) and (63), and in the process the  $\vec{k}$ -integration is done after the dispersive character of the surface waves is incorporated explicitly up to linear terms in  $\vec{k}$  by the equation

$$\sigma_f = \sigma_B + (c_{gB} \vec{\sigma}(k - k_B)) \quad (69)$$

The  $\vec{k}$ -integration returns  $\delta$ -functions with arguments  $\vec{x}' - \vec{x}'' + \vec{c}_g(t' - t'') - \vec{U}' t' + \vec{U}'' t''$ , and, in principle these allow one of the  $\vec{x}$ -integrals to be performed. The result is

$$\begin{aligned}
 \langle I(\vec{x}_0) \rangle - \langle I(\vec{x}_0) \rangle_s &= 4\pi^2 P_s(\vec{k}_B) \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} I^*(\vec{x}_1'', \vec{x}_0, t'') I(\vec{x}', \vec{x}_0, t') A_1^*(\vec{k}_B, \sigma_B; \vec{x}', t) \\
 &\times e^{-i(\sigma_B - c_{gB} \vec{k}_B)(t' - t'')} J(\vec{x}_1'', \vec{x}') d\vec{x}' dt' dt'' \\
 &+ 4\pi^2 P_s^*(-\vec{k}_B) \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} I^*(\vec{x}_2'', \vec{x}_0, t'') I(\vec{x}', \vec{x}_0, t') A_1(-\vec{k}_B, \sigma_B; \vec{x}', t') \\
 &\times e^{i(\sigma_B - c_{gB} \vec{k}_B)(t' - t'')} J(\vec{x}_2'', \vec{x}') d\vec{x}' dt' dt'' \quad (70)
 \end{aligned}$$

with

$$\vec{x}_{1,2}'' = \vec{x}' \pm \vec{c}_{gB}(t' - t'') - \vec{U}(\vec{x}', t')t' + \vec{U}(\vec{x}_{1,2}'', t'')t'' \quad (71)$$

implicitly defining  $\vec{x}_{1,2}''$  ( $\vec{x}_1''$  for upper sign and  $\vec{x}_2''$  for lower sign). In Equation (70),  $J$  is the Jacobian allowing the  $\delta$ -functions to be integrated and it has the form

$$J = (1 + t'' \frac{\partial U_x}{\partial x''}) (1 + t'' \frac{\partial U_y}{\partial y''}) \quad (72)$$

evaluated at  $\vec{x}_1''$  or  $\vec{x}_2''$  as appropriate.

The modulation  $A_1$  is usually described in terms of the energy envelope  $E(\vec{k}, \sigma_f; \vec{x}', t')$  in which case, by defining the autocorrelation function of the  $\varsigma$ -process in terms of the Fourier transform of  $P_s(\vec{k})E$  and equating that description to the equivalent expression from Equation (58), the following result pertains:



$$|A(\vec{k}, \sigma_f; \vec{x}', t')|^2 = (E + E^*)/2 \quad (73)$$

or, in terms of  $A_1$  and an equivalent  $E_1$  defined as  $E-1$ ,

$$\begin{aligned} |A|^2 &= 1 + A_1 + A_1^* + \dots \\ &= 1 + (E_1 + E_1^*)/2 \end{aligned} \quad (74)$$

and so,

$$A_1 \approx E_1/2 \quad (75)$$

For usual applications, current strain rates<sup>11</sup> are less than  $0(10^{-2} \text{s}^{-1})$  and SAR processing times<sup>2</sup> are less than 10 s, thus in Equations (72) and (70),  $J$  can be approximated quite well simply as unity. This also simplifies Equation (71) to the explicit form

$$\vec{x}_{1,2}'' = \vec{x}' + \vec{c}_{gB}(t' - t'') - \vec{U}(\vec{x}', t')(t' - t'') \quad (76)$$

The propagational character of the internal waves can also be incorporated explicitly, and here for simplicity, only dispersionless internal waves will be considered. A single transformation can be made from  $\vec{x}', t'$  and  $\vec{x}'', t''$  frames to coordinates moving with the internal wave  $\vec{\eta}, t'$  and  $\vec{\eta}_{1,2}, t''$ :

$$\vec{x}' = \vec{\eta} + \vec{c}_I t' \quad (77a)$$

$$\vec{x}_{1,2}'' = \vec{\eta}_{1,2} + \vec{c}_I t'' \quad (77b)$$

where  $\vec{c}_I$  is the phase (and group) velocity of the internal waves. With these changes, Equation (70) can be rewritten as follows (note, from Equation (62),  $P_s$  is real):

$$\begin{aligned}
\langle I(\vec{x}_0) \rangle - \langle I(\vec{x}_0) \rangle_s = & 2\pi^2 P_S(\vec{k}_B) \int_{-\infty}^{\infty} E_1^*(\vec{k}_B, \sigma_f, \vec{\eta}) \int_{-\infty}^{\infty} I(\vec{\eta} + \vec{c}_I t', \vec{x}_0, t') \\
& \times I^*(\vec{\eta}_1 + \vec{c}_I t'', \vec{x}_0, t'') e^{-i(\sigma_B - c_{gB} \vec{k}_B)(t' - t'')} dt' dt'' d\vec{\eta} \\
& + 2\pi^2 P_S(-\vec{k}_B) \int_{-\infty}^{\infty} E_1(-\vec{k}_B, \sigma_f, \vec{\eta}) \int_{-\infty}^{\infty} I(\vec{\eta} + \vec{c}_I t', \vec{x}_0, t') \\
& \times I^*(\vec{\eta}_2 + \vec{c}_I t'', \vec{x}_0, t'') e^{i(\sigma_B - c_{gB} \vec{k}_B)(t' - t'')} dt' dt'' d\vec{\eta} \\
& + \text{complex conjugate}
\end{aligned} \tag{78}$$

With a further definition of the inner double integrals in Equation (78) as impulse response functions  $K_{a,b}$ , this last form becomes

$$\begin{aligned}
\langle I(\vec{x}_0) \rangle - \langle I(\vec{x}_0) \rangle_s = & 2\pi^2 P_S(\vec{k}_B) \int_{-\infty}^{\infty} E_1(\vec{k}_B, \sigma_f, \vec{\eta}) K_a(\vec{\eta}, \vec{x}_0) d\vec{\eta} \\
& + 2\pi^2 P_S(-\vec{k}_B) \int_{-\infty}^{\infty} E_1^*(-\vec{k}_B, \sigma_f, \vec{\eta}) K_b(\vec{\eta}, \vec{x}_0) d\vec{\eta} \\
& + \text{complex conjugate}
\end{aligned} \tag{79}$$

where

$$\begin{aligned}
K_{a,b}(\vec{\eta}, \vec{x}_0) = & \int_{-\infty}^{\infty} I(\vec{\eta} + \vec{c}_I t', \vec{x}_0, t') I^*(\vec{\eta}_{1,2} + \vec{c}_I t'', \vec{x}_0, t'') \\
& \times e^{\pm i(\sigma_B - c_{gB} \vec{k}_B)(t' - t'')} dt' dt''
\end{aligned} \tag{80}$$

with  $K_a$  taking the upper sign and  $\vec{\eta}_1$ , and  $K_b$  the lower sign and  $\vec{\eta}_2$ .

From Equation (79) it can be seen that the modulation in the image is simply a scaled, filtered version of the Bragg scatterer energy density added to a similar scaled filtered version of the anti-Bragg energy density. The kernel functions  $K_{a,b}$  represent filters that depend on position  $\vec{x}_0$  (and are not just functions of  $\vec{\eta}-\vec{x}_0$ ).

For the Gaussian beam case,  $I$  is given by Equation (24), and, using Equations (25) to (27) and after expanding the exponent in  $I$  to second order in  $t'$  and  $t''$ , the integrals in Equation (80) can be evaluated. The results can be put into a standard normal form as follows:

$$K_{a,b}(\vec{\eta}, \vec{x}_0) = \frac{z^4 \omega_c^4 T_G^2 Y_G^2}{64 \pi^2 R_0^8 c^4 \Delta^2 V_1^2} \Gamma_{a,b} \exp[-4 \ln 2 \left( \left( \frac{\eta_x - x_s}{\text{Res}_x} \right)^2 + \left( \frac{\eta_y - y_{sa,b}}{\text{Res}_y} \right)^2 \right) - B_{a,b}(\eta_x - x_s)(\eta_y - y_{sa,b})] \quad (81)$$

Here, the  $\text{Res}_{x,y}$ -factors are full-width at half-maximum, and the various terms to first order in  $c_B/V_1$ ,  $c_I/V_1$ ,  $U/V_1$  and  $(V-V_1)/V_1$  are

$$\text{Res}_x = \frac{c R_0 T_G}{x_0 (\omega^2 T_G^4 - 1)^{1/2}} \sqrt{\ln 2} \quad (82)$$

$$\text{Res}_y = \frac{2 c R_0 \sqrt{\ln 2}}{Y_G \omega_c (1 + c^2 R_0^2 / Y_G^4 \omega_c^2)^{1/2}}$$

$$\times \left( 1 - \frac{c I_y}{2 V_1} + \frac{V - V_1}{2 V_1} + i \left( \frac{c I_y - c_B B_y - U_y}{V_1} \right) \frac{\omega_c Y_G^2}{c R_0} \right) \quad (83)$$

$$x_s = x_0 - y_0 c I_x / V_1 \quad (84)$$

$$y_{sa} = y_{sb} = y_0 + y_0 \left( \frac{V - V_1 - c I_y}{V_1} \right) + x_0 \frac{c_B B_x + U_x}{V_1 (1 + c^2 R_0^2 / Y_G^4 \omega_c^2)} \quad (85)$$

$$B_a - B_b = \frac{c_g B_x + U_x - c I_x}{V_1} \left( \frac{2i Y_G^2 \omega_c x_o^2}{R_o^3 T_G^2 c^3} (1 - \omega'^2 T_G^4) \right. \\ \left. + \frac{c_g B_x + U_x}{V_1} \frac{\omega^2 Y_G^2 z_o^2}{c^2 R_o^4} + \frac{c I_x}{V_1} \frac{2x_o^2}{R_o^2 T_G^2 c^2} (1 - \omega'^2 T_G^4) \right) \quad (86)$$

$$\Gamma_a - \Gamma_b = e \left( \frac{c_g B_y + c I_y + U_y}{2V_1} \right) + \frac{i \omega_c Y_G^2}{c R_o} \left( \frac{c_g B_y - c I_y + U_y}{2V_1} \right) - \frac{V - V_1}{V_1} \quad (87)$$

The differences between these equations and those of the previous section (see particularly Equations (75) and (52)) are due completely to the product form of Equation (73) and its convolutional nature.  $Res_x$  here is less than  $Res_x$  in the Kelvin wake description by a factor of about  $\sqrt{2}$ .

It should be especially noted that the  $\vec{\eta}$ -behaviour of  $K_{a,b}$  is not confined to the explicit  $\eta_x$ ,  $\eta_y$  terms appearing on the right-hand side of Equation (81). The terms  $Res_y$ ,  $y_{sa,b}$  and  $\Gamma_{a,b}$  also contain  $\vec{\eta}$ -behaviour through  $U_x$  and  $U_y$ . The traditional velocity bunching term  $x_o U_x / V_1$  appears in Equation (85) and it produces not only the usual misregistrations but it can also produce variations in image intensity if there are variations in  $U_x$  with  $\eta_y$ . This can be seen by considering  $U_x$  to be linear in  $\eta_y$  in the exponent of Equation (74) thus effectively changing the y-resolution. In symbolic form, and with  $c^2 R_o^2 / Y_G^4 \omega_c^2$  ignored compared to 1 (as is typical), and  $U'_{xo} = (\partial U_x / \partial y)_o$ ,

$$U_x \approx U_{xo} + \eta_y U'_{xo} \quad (88)$$

$$y_s \approx y_{so} + \frac{x_o \eta_y U'_{xo}}{V_1} \quad (89)$$

$$K_{a,b} \approx \frac{z^4 \omega_c^4 T_G^2 Y_G^2}{64 \pi^2 R_o^8 c^4 \Delta^2 V_1^2} \Gamma e^{-4 \ln 2 \left( \frac{\eta_x - x_s}{Res_x} \right)^2 - 4 \ln 2 \left( \frac{\eta_y [1 - x_o U'_{xo} / V_1] - y_{so}}{Res_y} \right)^2} \\ \times e^{-B(\eta_x - x_s) \left( \eta_y [1 - \frac{x_o U'_{xo}}{V_1}] - y_{so} \right)} \quad (90)$$

If  $E_1$  were independent of  $\vec{\eta}$ , the  $E_1$ -terms could be removed from the integrals in Equation (79) and the integrations performed using Equation (90). The result would be

$$\begin{aligned} \langle I(\vec{x}_0) \rangle - \langle I(\vec{x}_0) \rangle_s &= 2\pi^2 [P_s(\vec{k}_B) E_1(\vec{k}_B) + P_s(-\vec{k}_B) E_1^*(-\vec{k}_B)] \\ &\times \frac{z^4 \omega_c^4 T_G^2 Y_G^2}{64\pi^2 R_0^8 c^4 \Delta^2 V_1^2} \Gamma \frac{\pi \text{Res}_x \text{Res}_y}{\sqrt{1 - \frac{\text{Res}_x^2 \text{Res}_y^2 B^2}{4}}} \frac{1}{1 - \frac{x_0 U'_o}{V_1}} \end{aligned} \quad (91)$$

with the scene intensity showing modulations due to the last term on the right even though no mean modulations were present in the original wave field. The same analysis could have been performed on the  $\langle I(x_0) \rangle_s$  term showing that it was perturbed by the same right-hand term, and then  $E_1$  could have been set to zero for the uniform wave field condition. Either way the result would be that the gradient of  $U_x$  in  $y$  produces pseudo-modulations in standard SAR-processed imagery.

For the symmetric processor case,  $y_0=0$  and it can be seen from Equations (82) to (87) that the scene motion and aircraft mismatch affect only  $\text{Res}_y$ ,  $y_s$  and  $\Gamma$ , but  $\text{Res}_x$  and  $x_s$  are unaffected to this approximation. For typical parameters (e.g. Equations (54a) to (54m))

$$\frac{\omega_c Y_G^2}{c R_0} \gg 1 \quad (92)$$

$$\omega' T_G^2 \gg 1 \quad (93)$$

and so

$$\text{Res}_x \approx \frac{c R_0}{x_0 \omega' T_G} \sqrt{\ln 2} \quad (94)$$

$$\text{Res}_y \approx \frac{2cR_o\sqrt{\ln 2}}{Y_G\omega_c} \frac{1 + \frac{\omega_c^2 Y_G^4}{c^2 R_o^2} \left( \frac{cI_y - c_g B_y - U_y}{V_1} \right)^2}{1 - \frac{cI_y}{2V_1} + \frac{V-V_1}{2V_1} - i \frac{\omega_c Y_G^2}{cR_o} \left( \frac{cI_y - c_g B_y - U_y}{V_1} \right)} \quad (95)$$

$$B = \frac{c_g B_x + U_x}{V_1} \frac{\omega^2 Y_G^2 z_o^2}{c^2 R_o^4} - \frac{cI_x}{V_1} \frac{2x_o^2 \omega'^2 T_G^2}{R_o^2 c^2} - i \frac{c_g B_x + U_x - cI_x}{V_1} \frac{2Y_G^2 \omega_c x_o^2 \omega'^2 T_G^2}{R_o^3 c^3} \quad (96)$$

$$x_s = x_o \quad (97)$$

$$y_s = x_o \frac{c_g B_x + U_x}{V_1} \quad (98)$$

and  $\Gamma$  is unchanged. With  $\text{Res}_y$  written as in Equation (95), the exponent of  $K_{a,b}$  from Equation (81) can be seen to separate quite readily into a real and an imaginary part. The resultant real resolution in  $y$  is thus obtained from the exponential of the real part of the  $K_{a,b}$  exponent, multiplied by the cosine of the imaginary part. The resolution width occurs when this product equals  $1/2$ . In symbolic form, and with the  $K_{a,b}$  exponent defined to be  $\phi_r + i\phi_i$ , the resolution width is obtained from

$$e^{\phi_r} \cos \phi_i = 1/2 \quad (99)$$

By using Equations (90) and (95) at  $\eta_x = x_s$  and with some algebraic manipulations, Equation (99) provides the result

$$(\text{Res}_y)_{\text{Real}} = \sqrt{\frac{\theta(a/b)}{b}} \frac{2cR_o}{Y_G\omega_c} \left\{ \left( 1 - \frac{cI_y}{2V_1} + \frac{V-V_1}{2V_1} \right)^2 + \frac{\omega_c^2 Y_G^4}{c^2 R_o^2} \left( \frac{cI_y - c_g B_y - U_y}{V_1} \right)^2 \right\} \quad (100)$$

where

$$a = \left( 1 - \frac{cI_y}{2V_1} + \frac{V-V_1}{2V_1} \right)^2 - \frac{\omega_c^2 Y_G^4}{c^2 R_o^2} \left( \frac{cI_y - c_g B_y - U_y}{V_1} \right)^2, \quad (101)$$

$$b = \frac{2\omega_c Y_G^2}{cR_o} \left| \left( \frac{cI_y - c_g B_y - U_y}{V_1} \right) \left( 1 - \frac{cI_y}{2V_1} + \frac{V - V_1}{2V_1} \right) \right| \quad (102)$$

and  $\theta(a/b)$  is defined implicitly by (see Figure 8)

$$\frac{\ln(2\cos\theta)}{\theta} = a/b \quad (103)$$

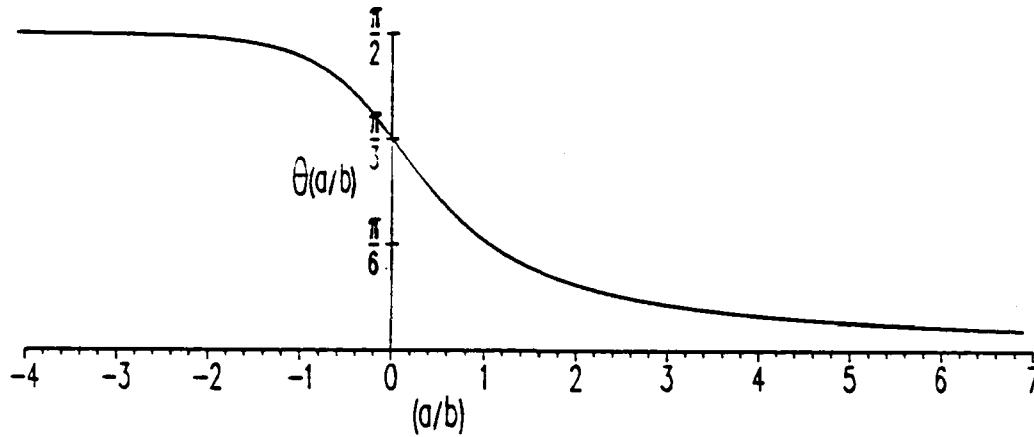


Figure 8. Behaviour of resolution defining function,  $\theta$ .

For

$$cI_y = c_g B = U_y = 0 \quad (104a)$$

and

$$V = V_1 \quad (104b)$$

i.e. no scene motion or aircraft mismatch effects,

$$a/b \rightarrow +\infty \quad (105a)$$

$$\theta \rightarrow \frac{b \ln 2}{a} \quad (105b)$$

$$a \rightarrow 1 \quad (105c)$$

and

$$(Res_y)_{real} \rightarrow \frac{2\sqrt{\ln 2} \ cR_0}{Y_G \omega_c} \quad (105d)$$

This is the traditional value and it decreases as the width of the processed beam,  $Y_G$ , increases.

For  $c_{Iy}$ ,  $c_{gBy}$ ,  $U$ ,  $\neq 0$  and  $Y_G \rightarrow \infty$ ,

$$a/b \rightarrow -\infty \quad (106a)$$

$$\theta \rightarrow \pi/2 \quad (106b)$$

and

$$(Res_y)_{real} \rightarrow Y_G^2 \sqrt{\frac{\pi \omega_c}{cR_0}} \left| \frac{c_{Iy} - c_{gBy} - U_y}{v_1} \right|^{3/2} \quad (106c)$$

In this limit the resolution increases as the beamwidth  $Y_G$  increases and it depends on the azimuthal translation speed of the surface wave Bragg scatterers ( $c_{gBy} + U_y$ ) relative to the azimuthal component of the internal wave phase velocity ( $c_{Iy}$ ).

For the numerical values of Equations (54a) to (54h), and with

$$c_{Iy} - (c_{gBy} + U_y) = 1 \text{ m/s} \quad (107)$$

the results from Equations (99) to (101) are

$$a/b = 0.4661 \quad (108a)$$

$$\theta = 0.785 \quad (108b)$$

and

$$(Res_y)_{real} \approx 4.19 \text{ m} \quad (109)$$



This can be compared with 3.16 m obtained from Equation (105d) showing that a not unreasonable amount of azimuthal scene motion can increase the resolution size by a significant amount (33% in this case).

To complete the description of the effect of scene motion it can be seen from Equation (98) that the velocity bunching term should actually be the range-component of the translational velocity of the Bragg-scatterers  $c_{gBx} + U_x$  rather than just  $U_x$  itself. Furthermore there are small amplitude and phase variations of the  $K$ 's due to the azimuthal components (and the processor mismatch) as is evident in the  $\Gamma$ 's (Equation (87)). The amplitude terms are only of order 1-10% and the phase term is of order 1/2 radian. There is also a small eccentricity of the impulse response peak as shown by the non-zero value of  $B_{a,b}$  in Equation (86), and for typical values this is of order 0.40 and dependent solely on the range-components of the scene velocities.

From Equation (59) the covariance of the image can be determined, and using the assumption of Gaussian statistics for the (narrow-band) transform coefficients  $p_s(\vec{k})$ , the covariance can be shown to be (see Appendix)

$$\begin{aligned} \text{cov}_I(\vec{\Delta}) = & 16\pi^4 P_s^2(\vec{k}_B) \left[ \left| \iint_{-\infty}^{\infty} (I_{P1}(\vec{x}, \vec{x}_0) I_{P1}^*(\vec{x}, \vec{x}_0 + \vec{\Delta}) + I_{P2}(\vec{x}, \vec{x}_0) I_{P2}^*(\vec{x}, \vec{x}_0 + \vec{\Delta})) d\vec{x} \right|^2 \right. \\ & \times \left. \left| \iint_{-\infty}^{\infty} (I_{P1}(\vec{x}, \vec{x}_0) I_{P2}(\vec{x}, \vec{x}_0 + \vec{\Delta}) + I_{P1}(\vec{x}, \vec{x}_0 + \vec{\Delta}) I_{P2}(\vec{x}, \vec{x}_0)) d\vec{x} \right|^2 \right] \quad (110) \end{aligned}$$

where  $I_{P1}$  is the unsquared impulse response function (i.e. whose width defines the pixel size of  $w(x_0)$ ) for positive  $\sigma_f$  and  $I_{P2}$  is the same for negative  $\sigma_f$ . It can be seen that if  $|\vec{\Delta}|$  is much larger than one pixel dimension the covariance is essentially zero because the products of the  $I_P$ 's are then zero.

## 5. Discussion

Kelvin wake narrow-V's in the SAR imagery are considered to be partly due to some randomization and intermittency in the wake structure itself, as is inferred from experimental evidence<sup>12</sup>, however, the peculiarities of the SAR processor described above do provide an alternative mechanism for enhancement of local regions of the wake. The fact that these are predicted to occur once per cycle of the transverse wave, and for the enhancements in each of the two arms of the narrow-V to occur at the same transverse wave phase angle, also lends credence to the validity of the effect, because these are in accordance with experimental data. No attempt has been made to model fully the Kelvin wake field, transverse and diverging, because it is known that accurate results for the short wavelength components are very difficult to achieve, requiring complicated source descriptions at the vicinity of the source ship, and also because of the intermittency just referred to. Indeed, the predicted SAR processor variations are not large enough by themselves, but combined with the intermittency and randomization of the direction of the diverging wavefield, they could possibly provide the mechanism underlying the appearance of the brightened V-arms.

The curvature effect examined in Section 3, Equations (50) to (53), can be much larger than traditional velocity bunching effects. The latter are known to produce radar cross section aberrations that are given by<sup>9</sup>

$$|\sigma_{VB}| = z_0 \left| \frac{\partial^2 \zeta_T}{\partial y^2} \right| \frac{c_{sy}}{V_1} \quad (111)$$

where the pattern is considered to be moving in the y-direction as a solid body. The comparable term from the curvature effect is

$$|\sigma_{cur}| = z_0 \left| \frac{\partial^2 \zeta_T}{\partial y^2} \right| 2 \left( \frac{\omega' T_G x_0}{\omega_c Y_G} \right)^2 \quad (112)$$

which is obtained by integrating  $K(\vec{\eta}, \vec{x}_0)$  over all  $\vec{\eta}$ , as if the Bragg-scatterers covered the entire pixel area uniformly, and using Equations (51), (52) and (53). In Equations (111) and (112) only the perturbations in  $\sigma$  are given. The denominator in  $\text{Res}_y$  from Equation (53) is also expanded binomially and the first non-unity term kept.

For the numerical values of Equations (54a-j), the relative magnitudes of  $\sigma_{\text{VB}}$  and  $\sigma_{\text{cur}}$  are 0.06 and 0.5, each of these values coming from the last terms in Equations (111) and (112) respectively. This evaluation is only for a very specific kind of surface (parabolic within the pixel area) and so is only indicative of possibilities. But it does show that curvature effects are not ignorable, a priori.

The perturbation in the SAR processor output due to the curvature of the corrugating wave can be understood from a simple geometrical picture of the radar wave at the sea surface. In the absence of the corrugation, a field of plane radar waves incident on the sea surface at an angle  $\theta_{\text{inc}}$  intersect the surface in a series of straight lines as shown in Figure 9(a). In the presence of a corrugation propagating perpendicularly to the radar waves (in the horizontal), the intersection of the radar waves at the surface also becomes corrugated, with local horizontal wavenumber directions that can be considerably deflected azimuthally from that of the main radar beam, Figure 9(b).

From Figure 10(a), it can be seen that, with  $z$  upwards, the radar crest that just intersects the sea surface, as shown, does so at

$$z_1 = (x_r - z \tan \theta_{\text{inc}}) \tan \theta_{\text{inc}} \quad (113)$$

If the surface height is specified by a long surface wave with height  $\varsigma$  given by

$$\varsigma = \varsigma_0 \cos(\vec{k} \cdot \vec{x} + \phi) \quad (114)$$

where  $\varsigma_0$  is the wave amplitude,  $\vec{k}$  is its wavenumber,  $\vec{x} = (x_r, x_a)$  are range and azimuth coordinates, and  $\phi$  is an arbitrary phase angle, then the intersection takes place at

$$z_1 = \varsigma \quad (115)$$

which, by Equations (113) and (114) is at

$$x_r - z \tan \theta_{inc} = \frac{s_0}{\tan \theta_{inc}} \cos(k \cdot x + \phi) \quad (116)$$

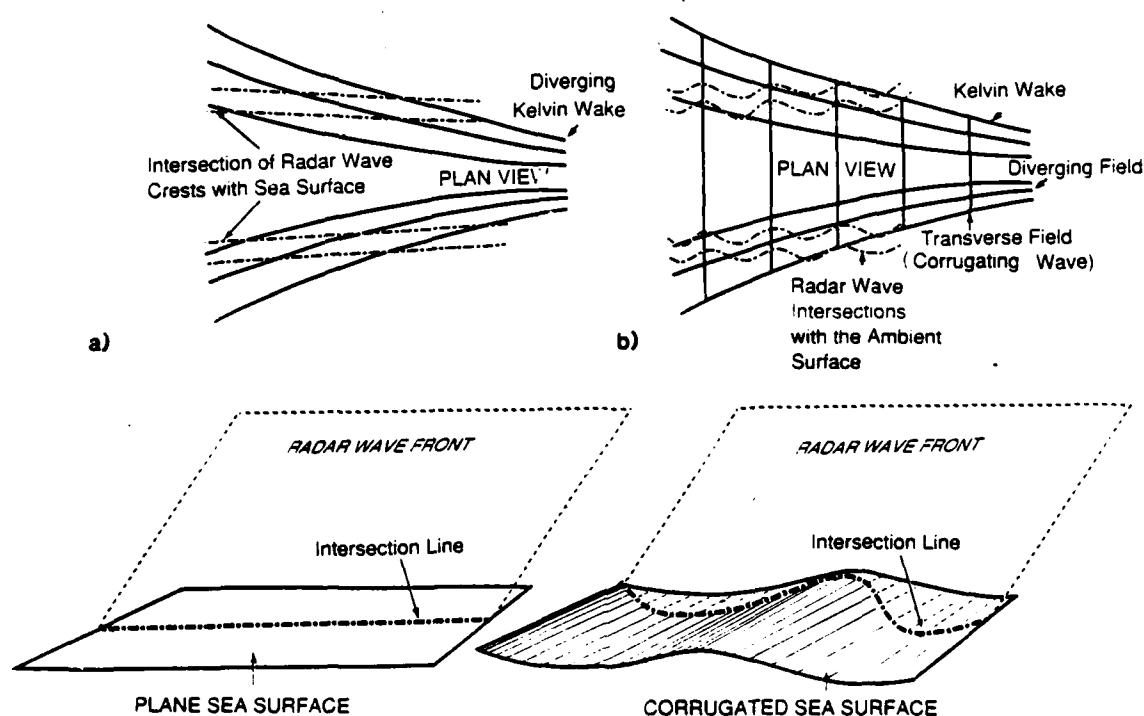


Figure 9. Intersection of the radar beam and the ocean surface (a) non-corrugated, (b) corrugated.

This last can be solved for  $x_r$  in terms of  $x_a$  (in principle) giving the horizontal shape of the intersection lines. For a corrugation propagating along the azimuth,  $k \cdot x = k_a x_a$  only, and

$$x_r = \frac{s_0}{\tan \theta_{inc}} \cos(k_a x_a - \phi) + z \tan \theta_{inc} \quad (117)$$

Typical plan view intersection lines are shown for this case in Figure 10(b).

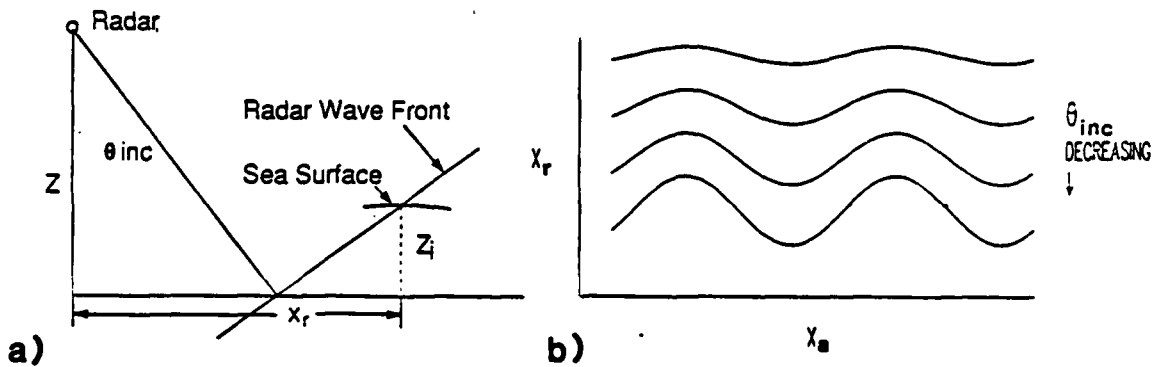


Figure 10. (a) Geometry of intersection (elevation view). (b) Typical horizontal intersection lines for a radar field in the presence of an azimuthally propagating corrugation (plan view).

The slopes of these lines provide estimates of the azimuthal deflection amounts for the effective radar beam in the Bragg scattering process. These deflections allow the scattering to take place from a wider range of horizontal surface wave propagation directions (in the azimuth) than would otherwise be possible. From Equation (117) the slopes are

$$\frac{dx_r}{dx_a} = - \frac{2\pi\zeta_0}{\lambda_s \tan \theta_{inc}} \sin\left(\frac{2\pi x_a}{\lambda_s} + \phi\right) \quad (118)$$

where  $\lambda_s$  is the wavelength. For the values of Equation (54a-m) but with  $\zeta_0 = 0.5$  m,

$$\left|\frac{dx_r}{dx_a}\right| = 0.177 = \tan 10^\circ \quad (119)$$

This is in accordance with the results of Section 3 where, from Figure 6, the centre of the SAR processor at  $\theta=0^\circ$  is aligned with waves propagating at  $9.6^\circ$  (at the maximum deviation).

The internal wave problem has been modelled by Fourier transforms and linear dispersion theory for the surface waves. This essentially resolves the wavefield into sinusoids of uniform height but of random amplitudes and phases (across different realizations). The measure of the coherence of the wave field is contained in its power spectrum and the bandwidths imposed by the radar configuration and the processor. In all cases, the surface waves imaged by the SAR/Bragg-scatter process are coherent over essentially one pixel dimension, regardless of the pixel size (for pixels large compared to a Bragg wavelength). This is apparent for the present model from Equation (110) and it follows directly from the fact that for larger pixels i.e. wider behaviour in  $l_p$ , better resolution is obtained in the Fourier domain, i.e. smaller bandwidths pertain and  $Q_2$  is narrower, but these result directly in larger coherence distances. In fact the two are quite simply complementary though the narrow-bandedness of the Fourier description of the power spectrum. For this particular model, broadband coherence distances are irrelevant; only the narrow-band coherences enter the problem. However, if strong non-linearities were present in the surface waves then this simple conclusion would need to be reexamined.

At each pixel location the mean image (Equation (60) rather than Equation (59)), because of its squared nature and all of the SAR processing, is due ultimately to the power in the Bragg waves covering the entire pixel area. If the Bragg wavelengths were much smaller than a pixel dimension so that individual scattering centers could be postulated within the pixel area, then internal correlations between Bragg waves resolved around their

individual centres could be established, and, in keeping with previous work,<sup>1</sup> would likely show uncorrelatedness. In that case the bandwidth of the scattering process would need to be large enough to allow that detailed structure to be faithfully captured in the scattered radar beam, and the SAR processor would need to be wide enough in bandwidth to accept it, otherwise the internal structure within the pixel would be smoothed out and some of the uncorrelatedness would disappear. When the bandwidth of the processor is due to the pixel dimensions, as is the case here (Equations (64) or A(6)), essentially no structure is allowed in the scattering waves within the pixel, and so internal decorrelation distances (or times) cease to be meaningful, and the signal reflected from the pixel becomes uniform across the pixel area, i.e. internally coherent. The power in this coherent signal, as stated earlier, is equal to the average power over the pixel area. This occurs because the Bragg-scattering process is essentially a Fourier transforming process with the SAR processor setting the bandwidths and these in turn setting the pixel size. It should be noted that this concept has been explored previously in connection with surface waves.<sup>2</sup>

The variation of  $Res_y$  with  $Y_G$  (Equation (99)) is highly significant in attempts to produce very small pixel SAR's to image internal waves. In the example shown in the last section, Equations (106) to (109), a processor that produces a  $Res_y$ -value of 3.16 m in the absence of azimuth velocities provides a value of 4.19 m in the presence of 1 m/s components overall (see Equation (107)). If  $Y_G$  is increased to 256 m so that the unperturbed pixel size decreases to 2.0 m, the effect of the azimuthal components is to increase this resolution width to 5.4 m, and in fact, further increases in  $Y_G$  result in even larger resolution widths. For these parameters an L-band  $Res_y$ -value of 2 m is possible but only if  $R_0$  is 560 km for  $Y_G$  at 162.2 m, and an even larger  $R_0$  for  $Y_G=256$  m.

For currents and internal waves moving in the x-direction only,  $Res_x$  and  $Res_y$  both take on their unperturbed values but the processor peak is rotated by  $B$ , and shifted by  $y_*$  (Equations (96), (98)). The  $B$ -value also

produces some eccentricity, and if the unperturbed processor  $Res_x$ - and  $Res_y$ -values are equal (as they are for the parameters of Section 3), the eccentricity is of order  $(U_x/V_1)^{1/2}$ , i.e.  $\sim 0.1$  and so not of great importance. Here,  $U_x$  is the x-component of the surface velocity, and eccentricity is defined as for an ellipse, i.e.  $\sqrt{a^2-b^2}/a$  where  $a$  and  $b$  are the semi-major axes respectively.

## 6. Conclusion

For the Kelvin wake problem the main conclusion is that large scale surface curvatures can exert a major influence on the processed SAR imagery. For particular cases, as examined in Section 3, the curvatures associated with the corrugating effect of the wake's transverse wave component can readily rotate the processor's Bragg acceptance angle from being bore-sighted with the radar beam to being  $\sim 10^\circ$  fore or aft of the beam. In modelling SAR imaging of Kelvin wakes, particularly for narrow-V effects, this mechanism should be included.

For internal waves, the main conclusion is that azimuthal propagational velocities can strongly affect the resultant azimuthal resolution width of the processor. Also, because of the Fourier description and the use of linear dispersion theory for surface waves, and 1<sup>st</sup>-order Bragg scattering for the radar reflection, the decorrelation distances are all of order one pixel size. Velocity bunching is shown to be caused by  $C_{gBx} + U_x$  rather than just  $U_x$ , where  $C_{gBx}$  is the range component of the Bragg wave group velocity and  $U_x$  is the range-component of the horizontal surface current (vertical velocities have been specifically excluded for this problem). Velocity bunching also produces pseudo-modulation in the imaging if  $\partial U_x / \partial y$  is non-zero, where  $y$  is the azimuth coordinate.



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## Appendix A

### Image Covariance

Equation (58) can be written as

$$w(\vec{x}_0) = \iint_{-\infty}^{\infty} p_s(\vec{k}) K_1(\vec{k}, \vec{x}_0) d\vec{k} + \iint_{-\infty}^{\infty} p_s^*(\vec{k}) K_2(\vec{k}, \vec{x}_0) d\vec{k} \quad (A1)$$

where  $K_1$  and  $K_2$  represent the triple integrals of the A- and I-functions. The covariance of the image is defined as

$$\text{cov}_I(\vec{\Delta}) = \langle I(\vec{x}_0 + \vec{\Delta}) I(\vec{x}_0) \rangle - \langle I(\vec{x}_0 + \vec{\Delta}) \rangle \langle I(\vec{x}_0) \rangle \quad (A2)$$

$$= \langle |w(\vec{x}_0 + \vec{\Delta})|^2 |w(\vec{x}_0)|^2 \rangle - \langle |w(\vec{x}_0 + \vec{\Delta})|^2 \rangle \langle |w(\vec{x}_0)|^2 \rangle \quad (A3)$$

The latter form of this equation can be written out in full using Equation (41), and the expectation operator when applied to the quadruple products of the  $p_s(\vec{k})$ 's can be seen to return double products of the power spectrum  $P_s(\vec{k})$ , or zero (using Gaussian statistics for  $p_s(\vec{k})$ ). After some lengthy algebraic analysis using Equations (63), (64) and (67), the result for  $\text{cov}_I$  becomes

$$\begin{aligned} \text{cov}_I(\vec{\Delta}) = & \left| \iint_{-\infty}^{\infty} P_s(\vec{k}) (K_1(\vec{k}, \vec{x}_0) K_1^*(\vec{k}, \vec{x}_0 + \vec{\Delta}) + K_2(\vec{k}, \vec{x}_0) K_2^*(\vec{k}, \vec{x}_0 + \vec{\Delta})) d\vec{k} \right|^2 \\ & + \left| \iint_{-\infty}^{\infty} P_s(\vec{k}) (K_1(\vec{k}, \vec{x}_0) K_2(\vec{k}, \vec{x}_0 + \vec{\Delta}) + K_1(\vec{k}, \vec{x}_0 + \vec{\Delta}) K_2(\vec{k}, \vec{x}_0)) d\vec{k} \right|^2, \quad (A4) \end{aligned}$$

and on keeping only the largest terms Equation (A4) becomes,

$$\begin{aligned}
 \text{cov}_I(\vec{\Delta}) = & \left| \iint_{-\infty}^{\infty} P_s(\vec{k}) (Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0) Q_2^*(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0 + \vec{\Delta}) \right. \\
 & + Q_2(\vec{k}, -\sigma_f, \vec{k}; \vec{x}_0) Q_2^*(\vec{k}, -\sigma_f, \vec{k}; \vec{x}_0 + \vec{\Delta}) \left. \right| d\vec{k} |^2 \\
 & + \left| \iint_{-\infty}^{\infty} P_s(\vec{k}) (Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0) Q_2(\vec{k}, -\sigma_f, \vec{k}; \vec{x}_0 + \vec{\Delta}) \right. \\
 & + Q_2(\vec{k}, \sigma_f, \vec{k}; \vec{x}_0 + \vec{\Delta}) Q_2(\vec{k}, -\sigma_f, \vec{k}; \vec{x}_0) \left. \right| d\vec{k} |^2
 \end{aligned} \tag{A5}$$

By using the argument given in the main text that  $Q_2$  is a narrow-band function of  $\vec{k}$ , it is possible to remove the  $P_s$ -function from the integrals. Also, the function  $Q_2$  is the Fourier transform of the spatial impulse response function of  $w(\vec{x}_0)$ , i.e. the smallest resolvable element in the imaged scene (unsquared), and this can be incorporated in the previous equation using the following definition

$$Q_2(\vec{k}, \pm\sigma_f, \vec{k}, \vec{x}_0) = \iint_{-\infty}^{\infty} e^{i\vec{k} \cdot \vec{x}} I_{p1,2}(\vec{x}, \vec{x}_0) d\vec{x} \tag{A6}$$

where  $+\sigma_f$  selects the impulse response function  $I_{p1}$  and  $-\sigma_f$  selects  $I_{p2}$ . With this definition, and the removal of the  $P_s(\vec{k})$ 's from the integration, Equation (A5) becomes Equation (110) of the main text.

The impulse response nature of  $I_{p1,2}$  can be seen from Equation (64) and the Fourier transform inverse to Equation (A6), which provide

$$I_{p1,2}(\vec{x}, \vec{x}_0) = \frac{1}{4\pi^2} \iiint_{-\infty}^{\infty} I(\vec{x}', \vec{x}_0, t') \iint_{-\infty}^{\infty} e^{i\vec{k} \cdot (\vec{x}' - \vec{x}) \pm i\sigma_f(\vec{k})t' - i\vec{k} \cdot \vec{U}' t'} d\vec{k} d\vec{x}' dt' \tag{A7}$$

and, by using Equation (69) the inner double integral can be performed, resulting in a  $\delta$ -function. By using the Jacobian argument leading up to Equation (76), the  $\delta$ -function can be integrated and the final result is

$$I_{p1,2}(\vec{x}, \vec{x}_0) = \int_{-\infty}^{\infty} I(\vec{x} \pm \vec{c}_{gB} t' + \vec{U}(\vec{x}) t', \vec{x}_0, t') e^{\mp i(\sigma_B - c_{gB} \vec{k}_B) t'} dt' \quad (A8)$$

These can be identified directly from Equation (21) as impulse response functions of  $w(x_0)$ , if the impulse propagates with velocity  $\pm \vec{c}_{gB} + \vec{U}$  and displays phase shifts of  $\pm(\sigma_B - c_{gB} \vec{k}_B) t'$ , i.e. if

$$s(\vec{x}', t') = \delta(\vec{x}' - [\vec{x} \pm \vec{c}_{gB} t' + \vec{U}(\vec{x}) t']) e^{\mp i(\sigma_B - c_{gB} \vec{k}_B) t'} \quad (A9)$$

This is precisely in accordance with the linear dispersive nature of the surface waves.

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